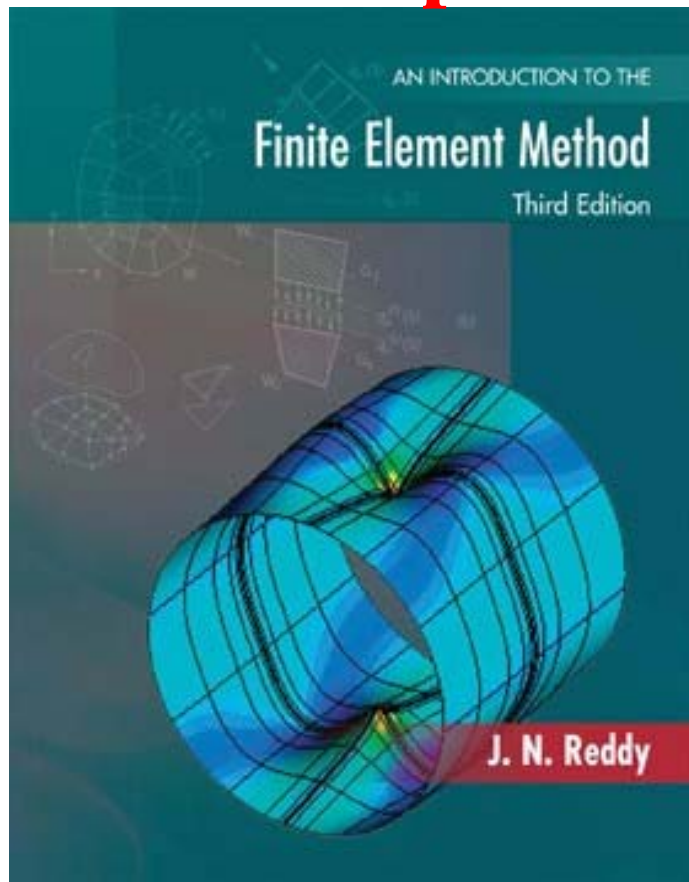


The Finite Element Method

Euler-Bernoulli and Timoshenko Beams

Read: Chapter 5



JN Reddy

CONTENTS

Euler-Bernoulli beam theory

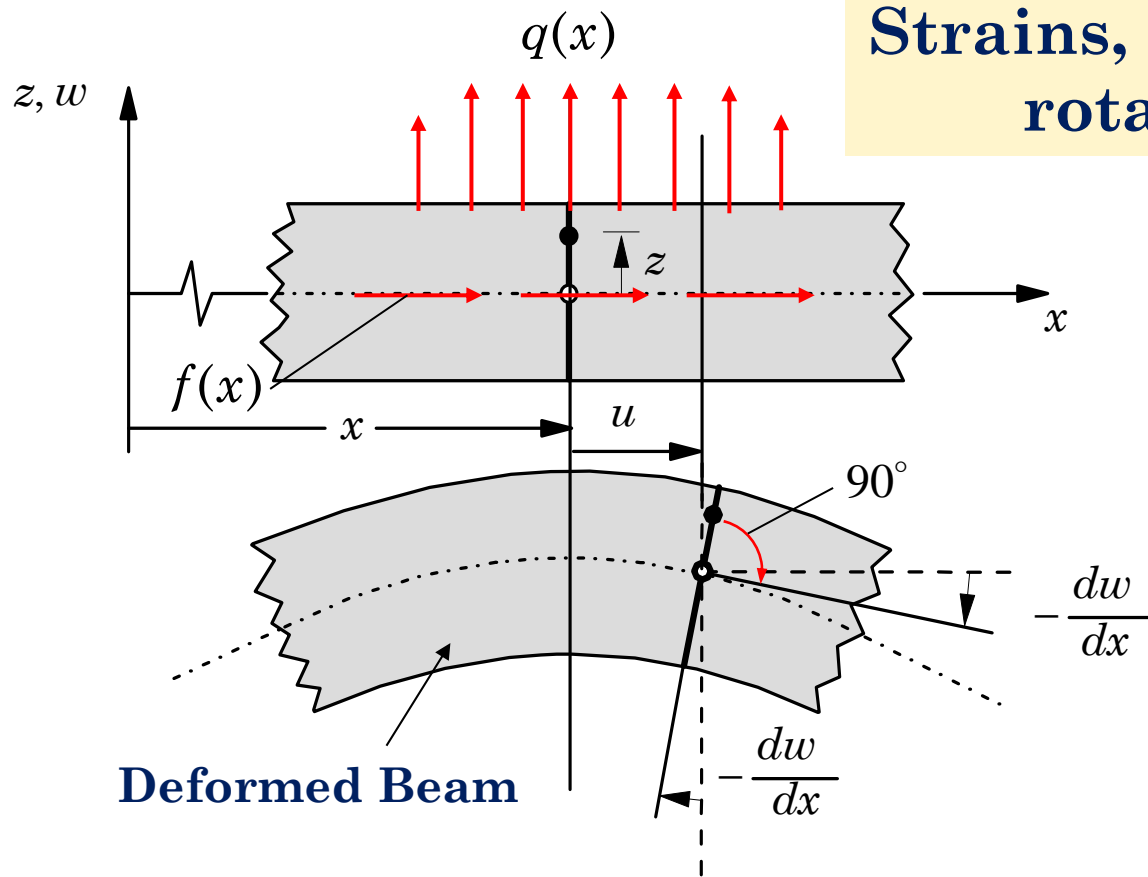
- Governing Equations
- Finite element model
- Numerical examples

Timoshenko beam theory

- Governing Equations
- Finite element model
- Shear locking
- Numerical example

KINEMATICS OF THE LINEARIZED EULER-BERNOULLI BEAM THEORY

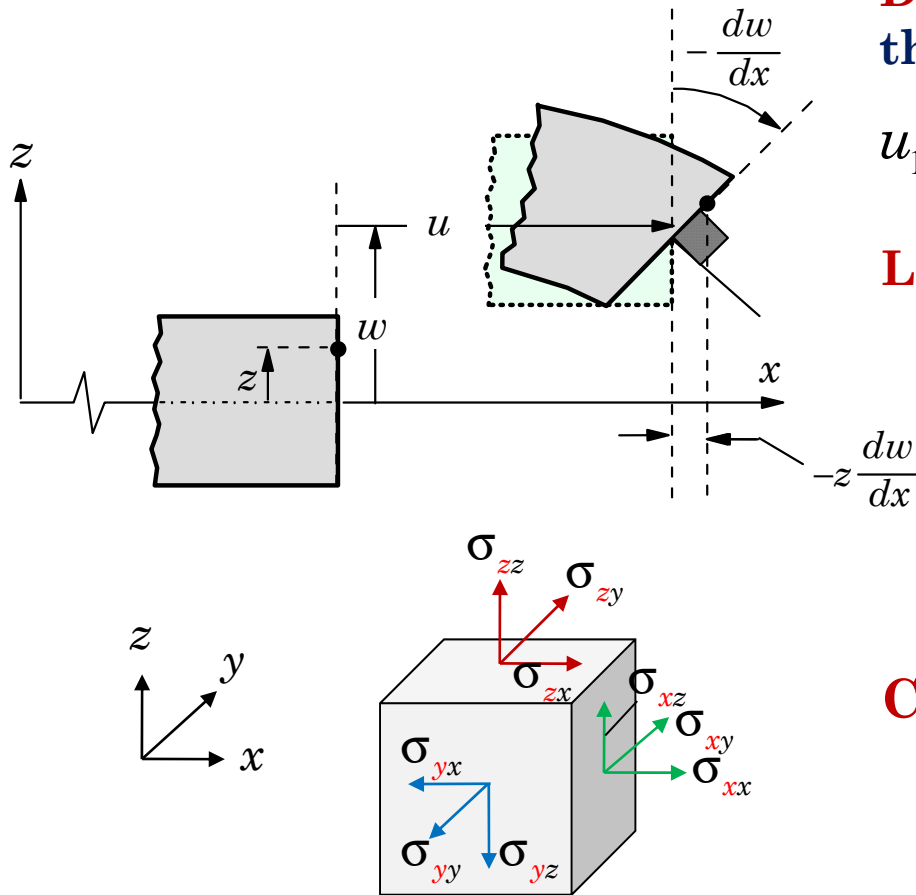
Strains, displacements, and rotations are small



Undeformed Beam

Euler-Bernoulli Beam Theory (EBT) is based on the assumptions of
(1) straightness,
(2) inextensibility, and
(3) normality

Kinematics of Deformation in the Euler-Bernoulli Beam Theory (EBT)



Displacement field (constructed using the hypothesis)

$$u_1(x, z) = u - z \frac{dw}{dx}, \quad u_2 = 0, \quad u_3 = w(x)$$

Linear strains

$$\epsilon_{11} = \epsilon_{xx} = \frac{\partial u_1}{\partial x_1} = \frac{du}{dx} - z \frac{d^2w}{dx^2},$$

$$\gamma_{xz} = \frac{\partial u_1}{\partial x_3} + \frac{\partial u_3}{\partial x_1} = -\frac{dw}{dx} + \frac{dw}{dx} = 0.$$

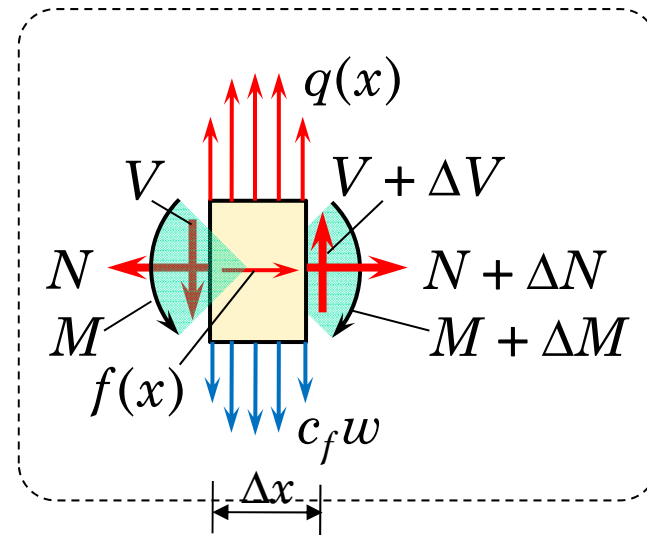
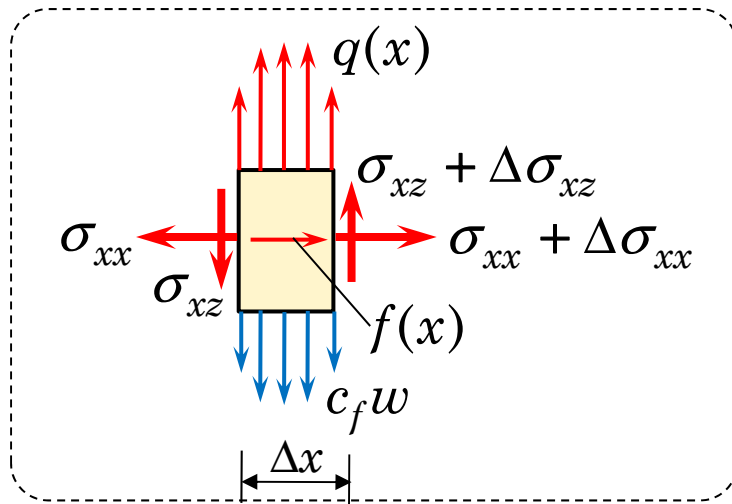
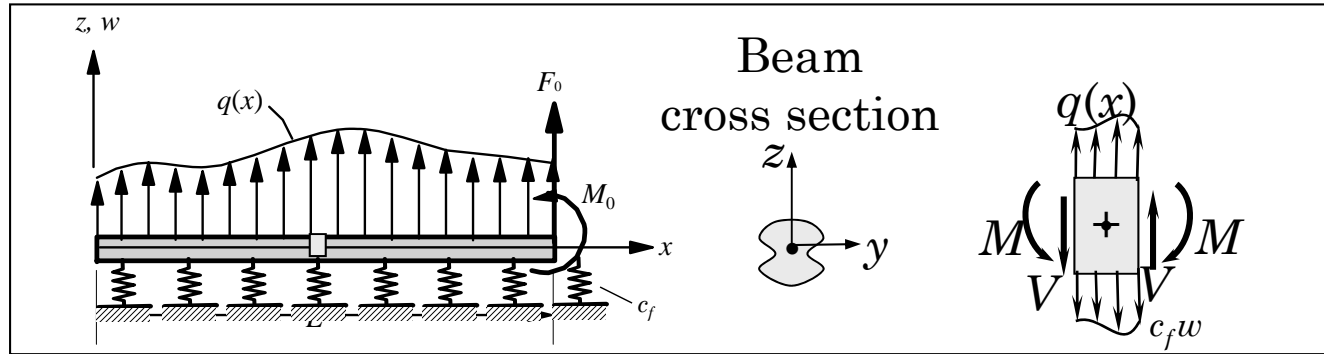
Constitutive relations

$$\sigma_{xx} = E \epsilon_{xx} = E \frac{du}{dx} - Ez \frac{d^2w}{dx^2},$$

$$\sigma_{xz} = G \gamma_{xz} = 0$$

Notation for stress components

Euler-Bernoulli Beam Theory



Definition of stress resultants

$$N = \int_A \sigma_{xx} dA, \quad M = \int_A \sigma_{xx} \cdot z dA, \quad V = \int_A \sigma_{xz} dA.$$



Euler-Bernoulli Beam Theory (Continued)

Equilibrium equations

$$\frac{dN}{dx} + f = 0, \quad \frac{dM}{dx} - V = 0, \quad \frac{dV}{dx} + q - c_f w = 0$$

$$\int_A 1 \cdot dA = A, \quad \int_A z \cdot dA = 0, \quad \int_A z^2 \cdot dA = I$$

Stress resultants in terms of deflection

$$N = \int_A \sigma_{xx} dA = \int_A \left(E \frac{du}{dx} - Ez \frac{d^2w}{dx^2} \right) dA = EA \frac{du}{dx}$$

$$M = \int_A \sigma_{xx} \times z dA = \int_A \left(E \frac{du}{dx} - Ez \frac{d^2w}{dx^2} \right) z dA = -EI \frac{d^2w}{dx^2}$$

$$V = \frac{dM}{dx} = \frac{d}{dx} \left(-EI \frac{d^2w}{dx^2} \right)$$

Euler-Bernoulli Beam Theory (Continued)

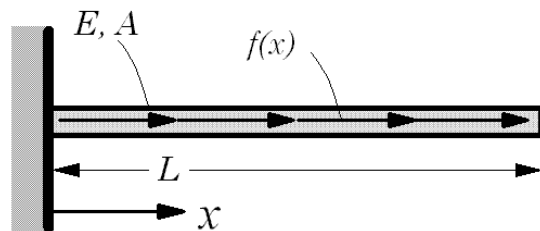
Governing equations in terms of the displacements

$$-\frac{d}{dx} \left(EA \frac{du}{dx} \right) - f = 0, \quad 0 < x < L$$

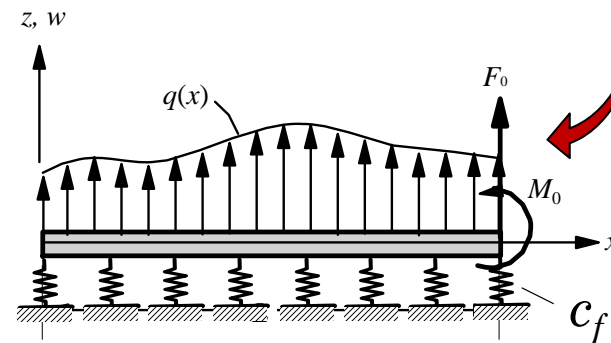
Bars
 u

$$\frac{d^2}{dx^2} \left(EI \frac{d^2 w}{dx^2} \right) + c_f w - q = 0, \quad 0 < x < L$$

Beams
 w



Axial deformation of a bar



Bending of a beam

Axial displacement is uncoupled from transverse displacement

Weak Form of the EB Beam Theory

Governing equation $\frac{d^2}{dx^2} \left(EI \frac{d^2 w}{dx^2} \right) + c_f w - q = 0, \quad 0 < x < L$

Weak form $\{v_i\}$ – set of weight functions

$$0 = \int_{x_a}^{x_b} v_i \left[\frac{d^2}{dx^2} \left(EI \frac{d^2 w_h}{dx^2} \right) + c_f w_h - q \right] dx$$

$$= \int_{x_a}^{x_b} \left[-\frac{dv_i}{dx} \frac{d}{dx} \left(EI \frac{d^2 w_h}{dx^2} \right) + c_f v_i w_h - v_i q \right] dx + \left[v_i \frac{d}{dx} \left(EI \frac{d^2 w_h}{dx^2} \right) \right]_{x_a}^{x_b}$$

Implies that the primary variable is w
(displacement)

Secondary variable
(shear force)

$$0 = \int_{x_a}^{x_b} \left[-\frac{dv}{dx} \frac{d}{dx} \left(EI \frac{d^2 w}{dx^2} \right) + c_f v w - v q \right] dx - v(x_a) Q_1 - v(x_b) Q_3$$

Weak Form (Continued)

$$0 = \int_{x_a}^{x_b} \left[EI \frac{d^2 v_i}{dx^2} \frac{d^2 w_h}{dx^2} + c_f v_i w_h - v_i q \right] dx - v_i(x_a) Q_1 - v_i(x_b) Q_3$$

Primary Variable, θ
Slope/rotation

$$+ \left[\left(-\frac{dv_i}{dx} \right) \cdot EI \frac{d^2 w_h}{dx^2} \right]_{x_a}^{x_b}$$

Secondary variable
(Bending Moment)

$$0 = \int_{x_a}^{x_b} \left[EI \frac{d^2 v_i}{dx^2} \frac{d^2 w_h}{dx^2} + c_f v_i w_h - v_i q \right] dx - v_i(x_a) Q_1 - v_i(x_b) Q_3$$

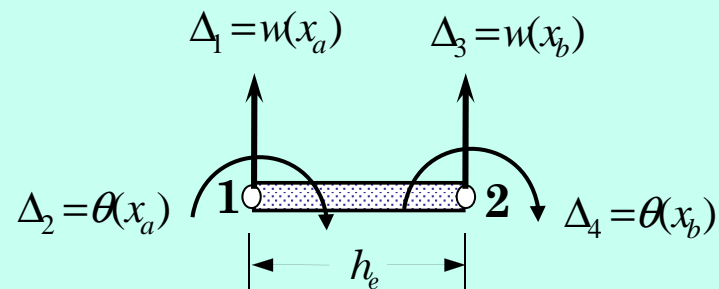
$$- \left(-\frac{dv_i}{dx} \right)_{x_a} \cdot Q_2 - \left(-\frac{dv_i}{dx} \right)_{x_b} \cdot Q_4$$

$$Q_1 = \left[\frac{d}{dx} \left(EI \frac{d^2 w_h}{dx^2} \right) \right]_{x_a} = -V_h(x_a), \quad Q_3 = \left[-\frac{d}{dx} \left(EI \frac{d^2 w_h}{dx^2} \right) \right]_{x_b} = V_h(x_b)$$

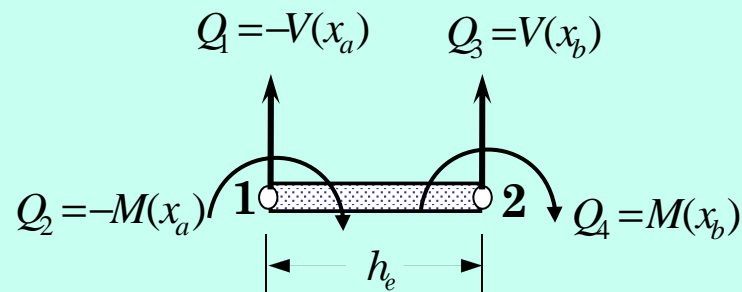
$$Q_2 = \left(EI \frac{d^2 w_h}{dx^2} \right)_{x_a} = -M_h(x_a), \quad Q_4 = \left(-EI \frac{d^2 w_h}{dx^2} \right)_{x_b} = M_h(x_b)$$

Beam Element Degrees of Freedom

Generalized displacements



Generalized forces





FINITE ELEMENT APPROXIMATION: Some Remarks

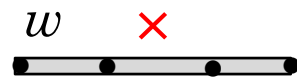
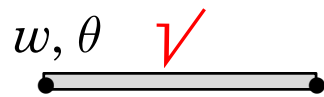
- **Continuity requirement based on the weak form, which requires that the second derivative of w exists and square-integrable.**
- **Continuity based on the primary variables, which requires carrying w and its first derivative as the nodal variables, requires cubic approximation w .**
- **Post-computation of secondary variables (bending moment and shear force) requires the third derivative of w to exist.**

FINITE ELEMENT APPROXIMATION

Primary variables (serve as the nodal variables that must be continuous across elements)

$$w, \theta = -\frac{dw}{dx}$$

$$w(x) \approx c_0 + c_1x + c_2x^2 + c_3x^3$$



Hermite cubic polynomials

$$\phi_1^e = 1 - 3 \left(\frac{x - x_a}{h_e} \right)^2 + 2 \left(\frac{x - x_a}{h_e} \right)^3$$

$$\phi_2^e = -(x - x_a) \left(1 - \frac{x - x_a}{h_e} \right)^2$$

$$\phi_3^e = 3 \left(\frac{x - x_a}{h_e} \right)^2 - 2 \left(\frac{x - x_a}{h_e} \right)^3$$

$$\phi_4^e = -(x - x_a) \left[\left(\frac{x - x_a}{h_e} \right)^2 - \frac{x - x_a}{h_e} \right]$$

$$w(x_a) \approx c_0 + c_1x_a + c_2x_a^2 + c_3x_a^3 \equiv \Delta_1$$

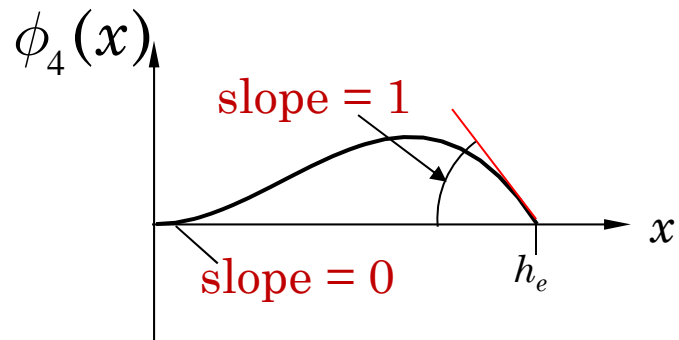
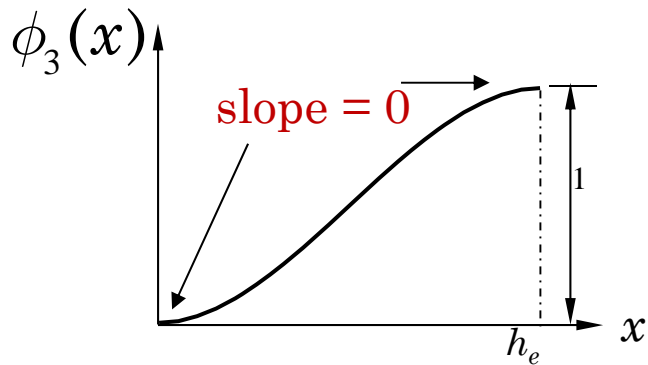
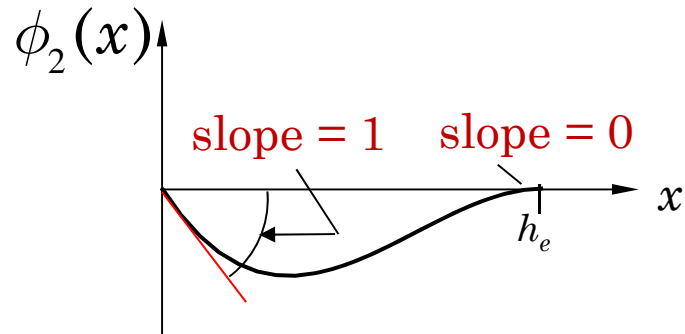
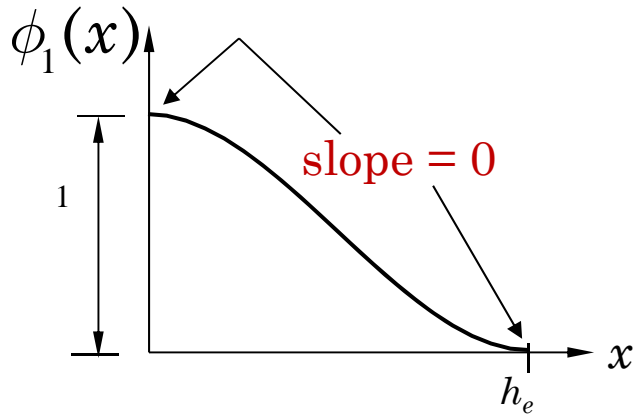
$$w(x_b) \approx c_0 + c_1x_b + c_2x_b^2 + c_3x_b^3 \equiv \Delta_3$$

$$\theta(x_a) \approx -c_1 - 2c_2x_a - 3c_3x_a^2 \equiv \Delta_2$$

$$\theta(x_b) \approx -c_1 - 2c_2x_b - 3c_3x_b^2 \equiv \Delta_4$$

$$w(x) \approx c_0 + c_1x + c_2x^2 + c_3x^3 = \sum_{j=1}^4 \Delta_j \phi_j(x)$$

HERMITE CUBIC INTERPOLATION FUNCTIONS $\phi_i(x)$



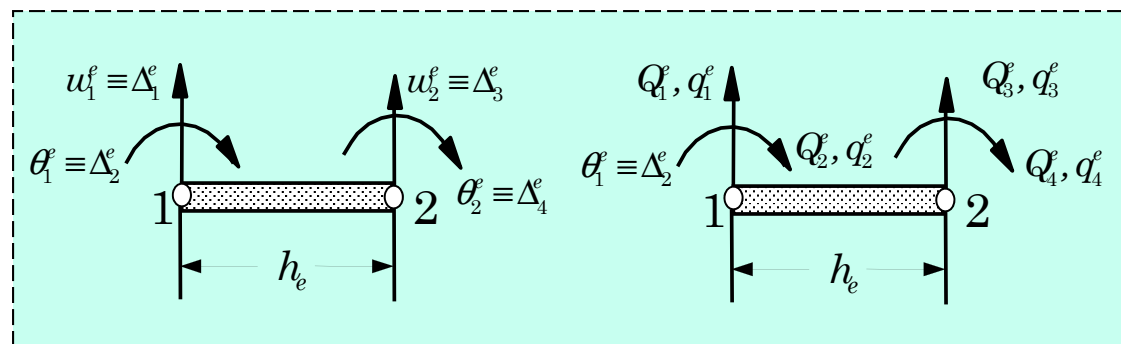
FINITE ELEMENT MODEL

$$w(x) \approx \sum_{j=1}^4 \Delta_j \phi_j(x)$$

$$\sum_{j=1}^4 K_{ij}^e \Delta_j^e - F_i^e = 0 \quad \text{or} \quad [K^e] \{\Delta^e\} = \{F^e\}$$

$$\begin{bmatrix} K_{11}^e & K_{12}^e & K_{13}^e & K_{14}^e \\ K_{21}^e & K_{22}^e & K_{23}^e & K_{24}^e \\ K_{31}^e & K_{32}^e & K_{33}^e & K_{34}^e \\ K_{41}^e & K_{42}^e & K_{43}^e & K_{44}^e \end{bmatrix} \begin{Bmatrix} \Delta_1^e \\ \Delta_2^e \\ \Delta_3^e \\ \Delta_4^e \end{Bmatrix} = \begin{Bmatrix} q_1^e \\ q_2^e \\ q_3^e \\ q_4^e \end{Bmatrix} + \begin{Bmatrix} Q_1^e \\ Q_2^e \\ Q_3^e \\ Q_4^e \end{Bmatrix}$$

$$K_{ij}^e = \int_{x_a}^{x_b} \left(EI \frac{d^2 \phi_i^e}{dx^2} \frac{d^2 \phi_j^e}{dx^2} + c_f \phi_i^e \phi_j^e \right) dx \quad F_i^e = \int_{x_a}^{x_b} \phi_i^e q \, dx + Q_i^e$$





Finite Element Model (Continued)

For element-wise constant values of $E_e I_e$ and q_e (and $c_f = 0$):

$$[K^e] = \frac{2E_e I_e}{h_e^3} \begin{bmatrix} 6 & -3h_e & -6 & -3h_e \\ -3h_e & 2h_e^2 & 3h_e & h_e^2 \\ -6 & 3h_e & 6 & 3h_e \\ -3h_e & h_e^2 & 3h_e & 2h_e^2 \end{bmatrix} \{F^e\} = \frac{q_e h_e}{12} \begin{Bmatrix} 6 \\ -h_e \\ 6 \\ h_e \end{Bmatrix} + \begin{Bmatrix} Q_1 \\ Q_2 \\ Q_3 \\ Q_4 \end{Bmatrix}$$

Postprocessing

$$M(x) = -EI \frac{d^2 w}{dx^2} = -EI \sum_{j=1}^4 \Delta_j^e \frac{d^2 \phi_j^e}{dx^2}$$

$$V(x) = \frac{dM}{dx} = -\frac{d}{dx} \left(EI \frac{d^2 w}{dx^2} \right) = -EI \sum_{j=1}^4 \Delta_j^e \frac{d^3 \phi_j^e}{dx^3}$$

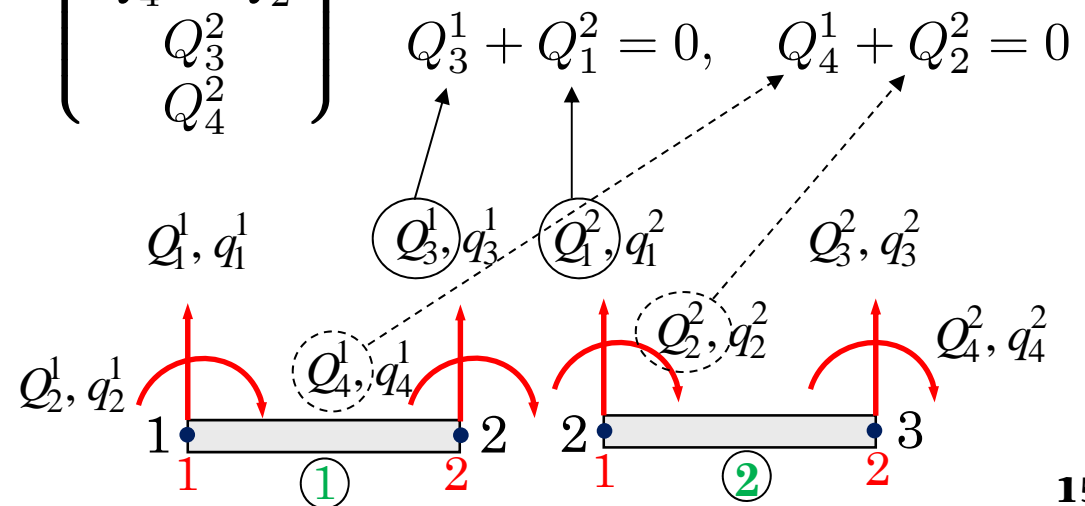
$$\sigma_x(x, z) = -\frac{M(x)z}{I} = Ez \frac{d^2 w}{dx^2} = Ez \sum_{j=1}^4 \Delta_j^e \frac{d^2 \phi_j^e(x)}{dx^2}$$

ASSEMBLY OF TWO BEAM ELEMENTS

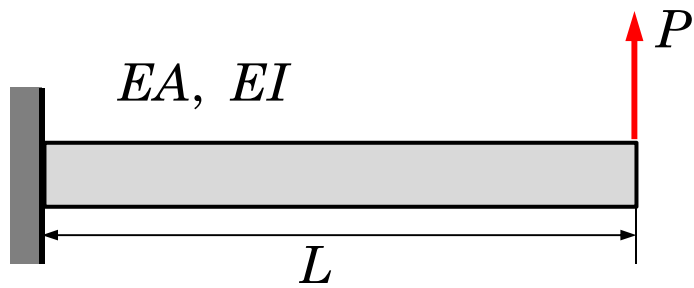
connected end-to-end

$$\frac{2EI}{h^3} \begin{bmatrix} 6 & -3h & -6 & -3h & 0 & 0 \\ -3h & 2h^2 & 3h & h^2 & 0 & 0 \\ -6 & 3h & 6+6 & 3h-3h & -6 & -3h \\ -3h & h^2 & 3h-3h & 2h^2+2h^2 & 3h & h^2 \\ 0 & 0 & -6 & 3h & 6 & 3h \\ 0 & 0 & -3h & h^2 & 3h & 2h^2 \end{bmatrix} \begin{Bmatrix} U_1 \\ U_2 \\ U_3 \\ U_4 \\ U_5 \\ U_6 \end{Bmatrix} \quad h = L/2$$

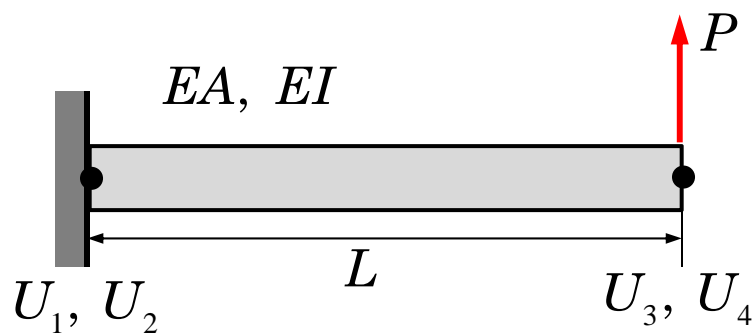
$$= \frac{q_0 L}{48} \begin{Bmatrix} 12 \\ -L \\ 24 \\ 0 \\ 12 \\ L \end{Bmatrix} + \begin{Bmatrix} Q_1^1 \\ Q_2^1 \\ Q_3^1 + Q_2^2 \\ Q_4^1 + Q_2^2 \\ Q_3^2 \\ Q_4^2 \end{Bmatrix}$$



A SIMPLE EXAMPLE - 1



Given problem



Boundary conditions:

$$U_1 = U_2 = 0, \quad Q_3 = P, \quad Q_4 = 0$$

$$[K^e] = \frac{2E_e I_e}{h_e^3} \begin{bmatrix} 6 & -3h_e & -6 & -3h_e \\ -3h_e & 2h_e^2 & 3h_e & h_e^2 \\ -6 & 3h_e & 6 & 3h_e \\ -3h_e & h_e^2 & 3h_e & 2h_e^2 \end{bmatrix}$$

Exact solution (according to the Euler-Bernoulli beam theory)

$$w(L) = \frac{PL^3}{3EI}$$

One element discretization

$$[K^e] \{\Delta^e\} = \{q^e\} + \{Q^e\} \quad h_e = L$$

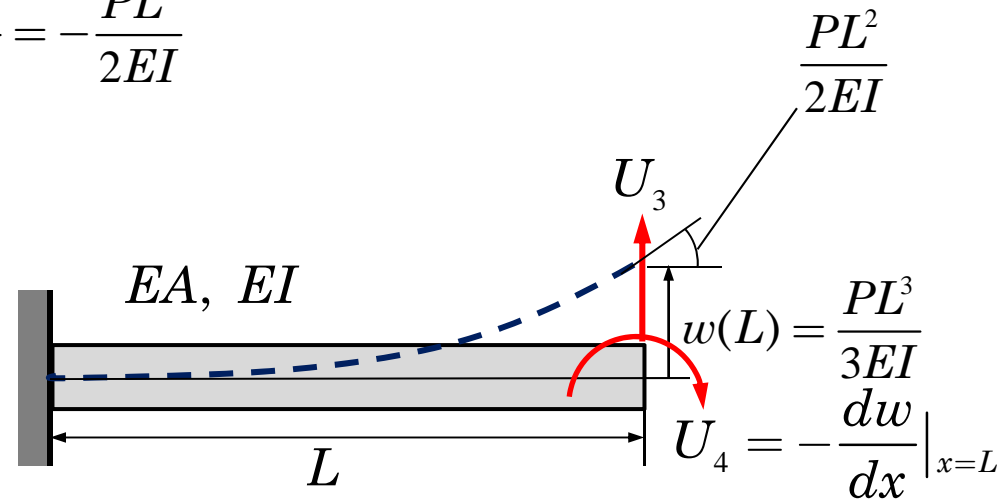
$$\{F^e\} = \frac{q_e h_e}{12} \begin{bmatrix} 6 \\ -h_e \\ 6 \\ h_e \end{bmatrix} + \begin{bmatrix} Q_1 \\ Q_2 \\ Q_3 \\ Q_4 \end{bmatrix}$$

A SIMPLE EXAMPLE – 1 (continued)

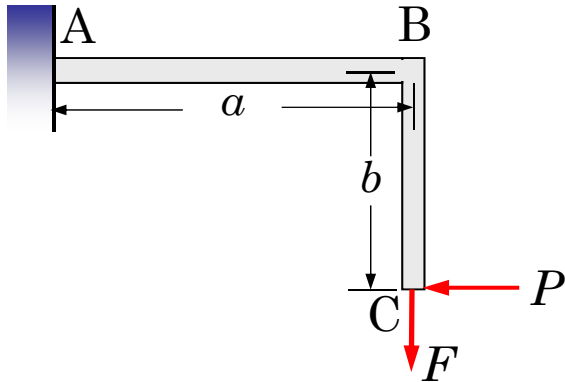
Solution using Cramer's rule

$$\begin{bmatrix} \frac{12EI}{L^3} & \frac{6EI}{L^2} \\ \frac{6EI}{L^2} & \frac{4EI}{L} \end{bmatrix} \begin{Bmatrix} U_3 \\ U_4 \end{Bmatrix} = \begin{Bmatrix} P \\ 0 \end{Bmatrix} \Rightarrow U_3 = \frac{\begin{vmatrix} P & \frac{6EI}{L^2} \\ 0 & \frac{4EI}{L} \end{vmatrix}}{\begin{vmatrix} \frac{12EI}{L^3} & \frac{6EI}{L^2} \\ \frac{6EI}{L^2} & \frac{4EI}{L} \end{vmatrix}} = \frac{(4PEI/L)}{[12(EI)^2/L^4]} = \frac{PL^3}{3EI}$$

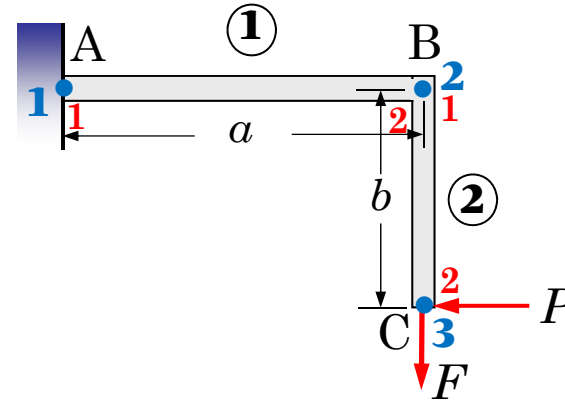
$$U_4 = \frac{\begin{vmatrix} \frac{12EI}{L^3} & P \\ \frac{6EI}{L^2} & 0 \end{vmatrix}}{\begin{vmatrix} \frac{12EI}{L^3} & \frac{6EI}{L^2} \\ \frac{6EI}{L^2} & \frac{4EI}{L} \end{vmatrix}} = \frac{-(6PEI/L^2)}{[12(EI)^2/L^4]} = -\frac{PL^2}{2EI}$$



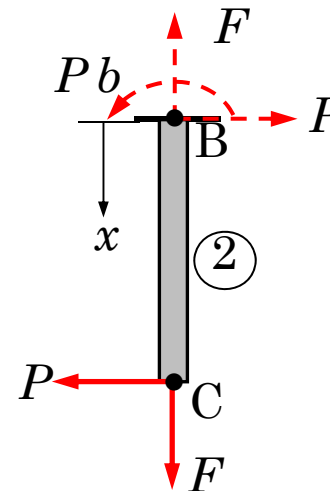
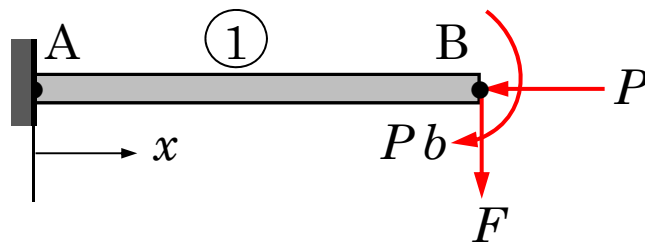
EXAMPLE - 2: A determinate frame structure



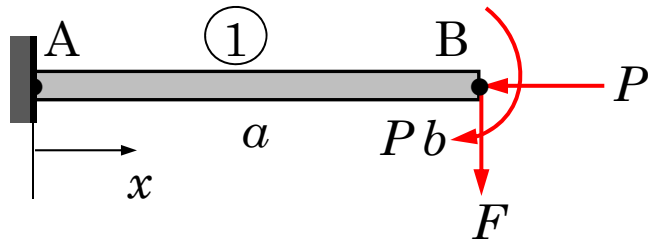
Given structure



Finite element discretization



EXAMPLE - 2 (continued)



Bar element, AB

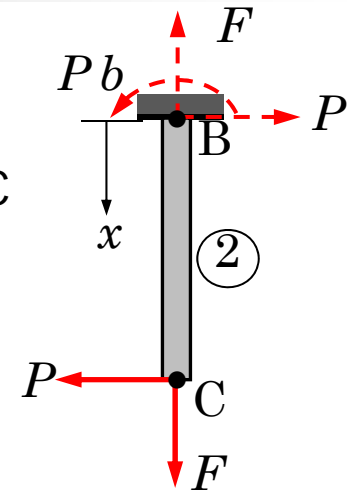
$$\frac{E_1 A_1}{a} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} u_A \\ u_B \end{Bmatrix} = \begin{Bmatrix} Q_A \\ Q_B \end{Bmatrix}$$

$$u_A = 0, Q_B = -P \Rightarrow u_B = -\frac{Pa}{E_1 A_1}$$

Beam element, AB

$$\frac{2E_1 I_1}{a^3} \begin{bmatrix} 6 & -3a & -6 & -3a \\ -3a & 2a^2 & 3a & a^2 \\ -6 & 3a & 6 & 3a \\ -3a & a^2 & 3a & 2a^2 \end{bmatrix} \begin{Bmatrix} w_A \\ \theta_A \\ w_B \\ \theta_B \end{Bmatrix} = \begin{Bmatrix} Q_1^A \\ Q_2^A \\ Q_1^B \\ Q_2^B \end{Bmatrix}$$

$$w_A = 0, \theta_A = 0, Q_1^B = -F, Q_2^B = Pb$$



Displacements at C relative to point B

Bar element, BC

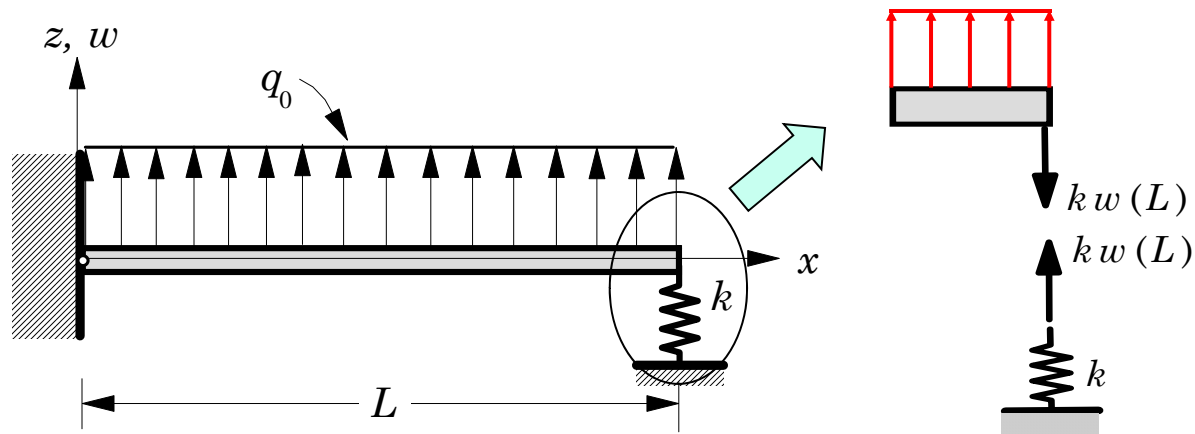
$$u_C = \frac{Fb}{E_2 A_2}$$

Beam element, BC

$$\frac{2E_2 I_2}{b^3} \begin{bmatrix} 6 & -3b & -6 & -3b \\ -3b & 2b^2 & 3b & b^2 \\ -6 & 3b & 6 & 3b \\ -3b & b^2 & 3b & 2b^2 \end{bmatrix} \begin{Bmatrix} w_B \\ \theta_B \\ w_C \\ \theta_C \end{Bmatrix} = \begin{Bmatrix} Q_1^B \\ Q_2^B \\ Q_1^C \\ Q_2^C \end{Bmatrix}$$

$$w_B = 0, \theta_B = 0, Q_1^C = -P, Q_2^C = 0$$

EXAMPLE – 3: Handling of a vertical spring



The free-body diagram of the beam element shows nodes 1 and 2. At node 1, there are forces Q_1^1 (up) and Q_2^1 (right). At node 2, there are forces Q_3^1 (up) and Q_4^1 (right). Displacements are $U_1 = U_2 = 0$ at node 1, and $U_3 \neq 0$ (up) and $U_4 = 0$ at node 2. A downward force $kw(L) = kU_3$ is applied at node 2. The relationship $Q_3^1 = -kw(L) = -kU_3$ is shown with a blue arrow.

Alternatively,

$$k \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} u_1^s \\ u_2^s \end{Bmatrix} = \begin{Bmatrix} Q_1^s \\ Q_2^s \end{Bmatrix}, \quad u_1^s = 0, u_2^s = U_3 \Rightarrow Q_2^s = kU_3$$

SOLUTION TO THE SPRING-SUPPORTED BEAM

$$\frac{2EI}{L^3} \begin{bmatrix} 6 & -3L & -6 & -3L \\ -3L & 2L^2 & 3L & L^2 \\ -6 & 3L & 6 & 3L \\ -3L & L^2 & 3L & 2L^2 \end{bmatrix} \begin{Bmatrix} U_1 = w_1 \\ U_2 = \theta_1 \\ U_3 = w_2 \\ U_4 = \theta_2 \end{Bmatrix} = \frac{q_0 L}{12} \begin{Bmatrix} 6 \\ -L \\ 6 \\ L \end{Bmatrix} + \begin{Bmatrix} Q_1 \\ Q_2 \\ Q_3 \\ Q_4 \end{Bmatrix} \begin{matrix} 0 \\ 0 \\ -kU_3 \\ 0 \end{matrix}$$

Boundary conditions

$$w_1 = 0, \theta_1 = 0, Q_3 = -kU_3, Q_4 = 0$$

Condensed equations for the unknown generalized nodal displacements

$$\begin{bmatrix} \frac{12EI}{L^3} + k & \frac{6EI}{L^2} \\ \frac{6EI}{L^2} & \frac{4EI}{L} \end{bmatrix} \begin{Bmatrix} U_3 \\ U_4 \end{Bmatrix} = \frac{q_0 L}{12} \begin{Bmatrix} 6 \\ -L \end{Bmatrix}$$

HANDLING OF A POINT SOURCES INSIDE AN ELEMENT

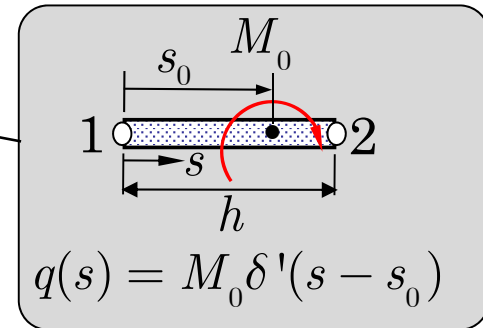
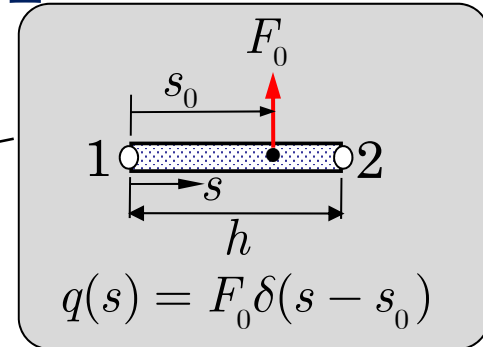
$$q_i = \int_0^h q(s) \phi_i(s) ds$$

$$q_i = \int_0^h q(s) \phi_i(s) ds = F_0 \phi_i(s_0), \quad i = 1, 2, 3, 4$$

$$q_i = \int_0^h q(s) \phi_i(s) ds = -M_0 \left. \frac{d\phi_i}{ds} \right|_{s=s_0}, \quad i = 1, 2, 3, 4$$

$$\phi_1(s) = 1 - 3\left(\frac{s}{h}\right)^2 + 2\left(\frac{s}{h}\right)^3, \quad \phi_2(s) = -s\left(1 - \frac{s}{h}\right)^2$$

$$\phi_3(s) = 3\left(\frac{s}{h}\right)^2 - 2\left(\frac{s}{h}\right)^3, \quad \phi_4(s) = -s\left[\left(\frac{s}{h}\right)^2 - \frac{s}{h}\right]$$



for F_0 placed
at $s_0 = 0.5h$

$$q_1 = \frac{F_0}{2}, \quad q_3 = \frac{F_0}{2}$$

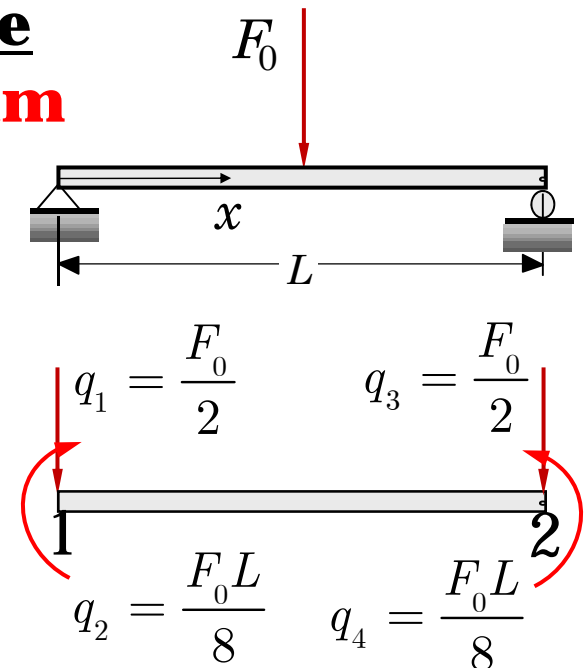
$$q_2 = -\frac{F_0 h}{8}, \quad q_4 = \frac{F_0 h}{8}$$

EXAMPLE - 4: A simply-supported beam

(a) Find the center deflection using one Euler-Bernoulli element in full beam

$$\frac{2EI}{L^3} \begin{bmatrix} 6 & -3L & -6 & -3L \\ -3L & 2L^2 & 3L & L^2 \\ -6 & 3L & 6 & 3L \\ -3L & L^2 & 3L & 2L^2 \end{bmatrix} \begin{Bmatrix} U_1 = w_1 \\ U_2 = \theta_1 \\ U_3 = w_2 \\ U_4 = \theta_2 \end{Bmatrix} = \begin{Bmatrix} q_1 \\ q_2 \\ q_3 \\ q_4 \end{Bmatrix} + \begin{Bmatrix} Q_1 \\ Q_2 \\ Q_3 \\ Q_4 \end{Bmatrix}$$

$-\frac{F_0 L}{8}$



Condensed equations

$$\frac{2EI}{L^3} \begin{bmatrix} 2L^2 & L^2 \\ L^2 & 2L^2 \end{bmatrix} \begin{Bmatrix} U_2 \\ U_4 \end{Bmatrix} = \frac{F_0 L}{8} \begin{Bmatrix} 1 \\ -1 \end{Bmatrix}$$

$$U_2 = \frac{F_0 L^2}{16EI}, \quad U_4 = -\frac{F_0 L^2}{16EI}$$

$$\begin{aligned} w(x) &= U_1 \phi_1(x) + U_2 \phi_2(x) + U_3 \phi_3(x) + U_4 \phi_4(x) \\ &= U_2 \phi_2(x) + U_4 \phi_4(x) \\ &= \frac{F_0 L^2}{16EI} \left\{ -x \left(1 - \frac{x}{L} \right)^2 + x \left[\left(\frac{x}{L} \right)^2 - \left(\frac{x}{L} \right) \right] \right\} \end{aligned}$$

$$w(0.5L) = \frac{F_0 L^2}{16EI} \left(-\frac{L}{8} - \frac{L}{8} \right) = -\frac{F_0 L^3}{64EI}$$

EXAMPLE - 4: A simply-supported beam

(b) Find the center deflection using one Euler-Bernoulli element in **half beam**

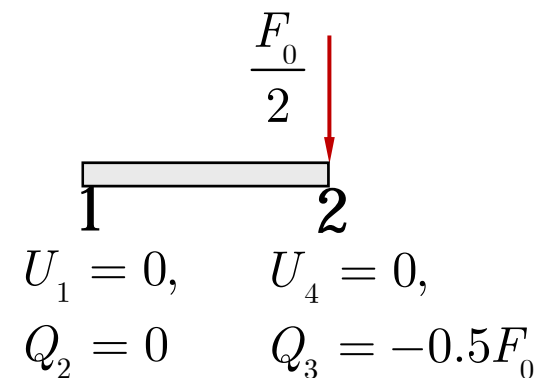
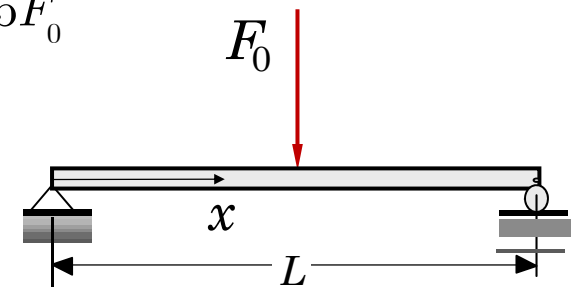
$$\frac{16EI}{L^3} \begin{bmatrix} 6 & -1.5L & -6 & -1.5L \\ -1.5L & 0.5L^2 & 1.5L & 0.25L^2 \\ -6 & 1.5L & 6 & 1.5L \\ -3L & 0.25L^2 & 1.5L & 0.5L^2 \end{bmatrix} \begin{Bmatrix} U_1 = w_1 \\ U_2 = \theta_1 \\ U_3 = w_2 \\ U_4 = \theta_2 \end{Bmatrix} = \begin{Bmatrix} Q_1 \\ Q_2 \\ Q_3 \\ Q_4 \end{Bmatrix} \begin{matrix} 0 \\ 0 \\ -0.5F_0 \\ 0 \end{matrix}$$

Condensed equations

$$\frac{16EI}{L^3} \begin{bmatrix} 0.5L^2 & 1.5L \\ 1.5L & 6 \end{bmatrix} \begin{Bmatrix} U_2 \\ U_3 \end{Bmatrix} = 0.5F_0 \begin{Bmatrix} 0 \\ -1 \end{Bmatrix}$$

$$U_2 = \frac{F_0 L^3}{32EI} \frac{4}{3L^2} \frac{1.5L}{1} = \frac{F_0 L^2}{16EI},$$

$$U_3 = \frac{F_0 L^3}{32EI} \frac{4}{3L^2} \frac{0.5L^2}{1} = \frac{F_0 L^3}{48EI}$$





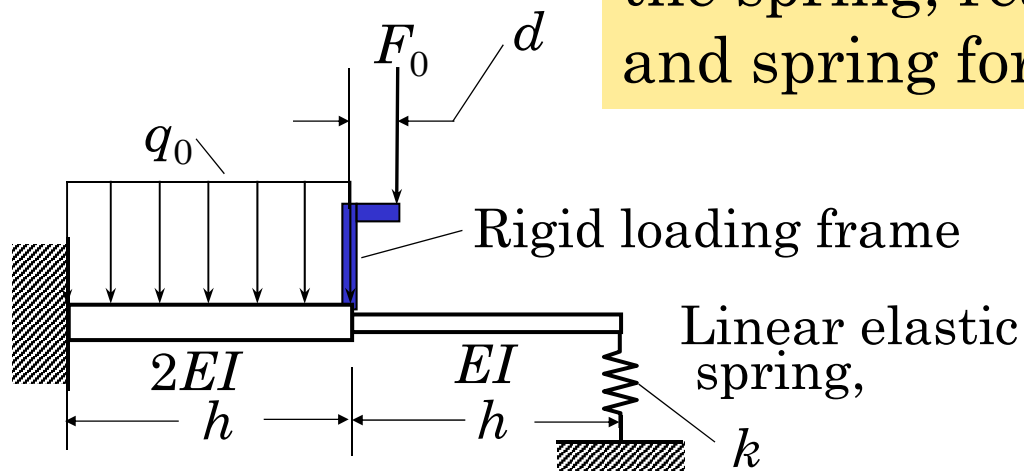
EXERCISE PROBLEM

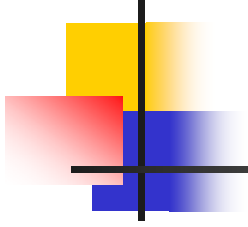
Problem: Develop weak form and the finite element model of the following equation, where w and P are unknowns:

$$\frac{d^2}{dx^2} \left(EI \frac{d^2 w}{dx^2} \right) + P \frac{d^2 w}{dx^2} = 0, \quad 0 < x < L$$

EXERCISE PROBLEM

Problem: Use the minimum number of EBT elements to find the compression in the spring, reactions at the fixed support, and spring force.

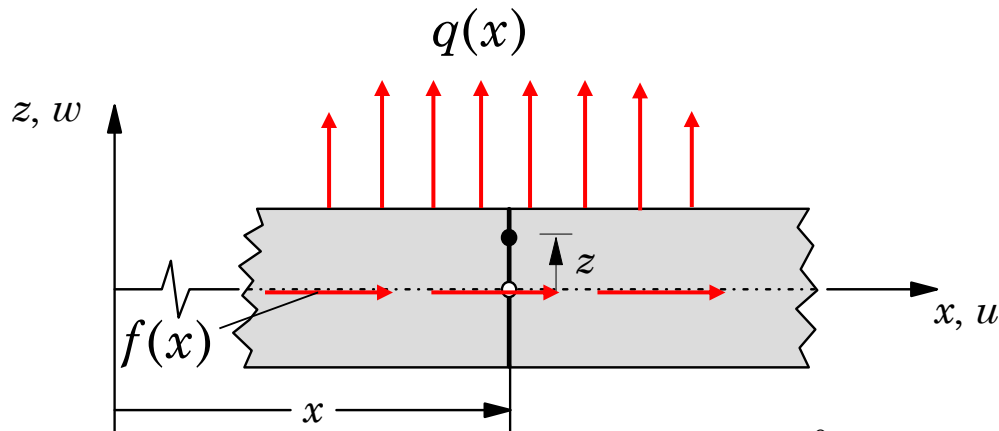




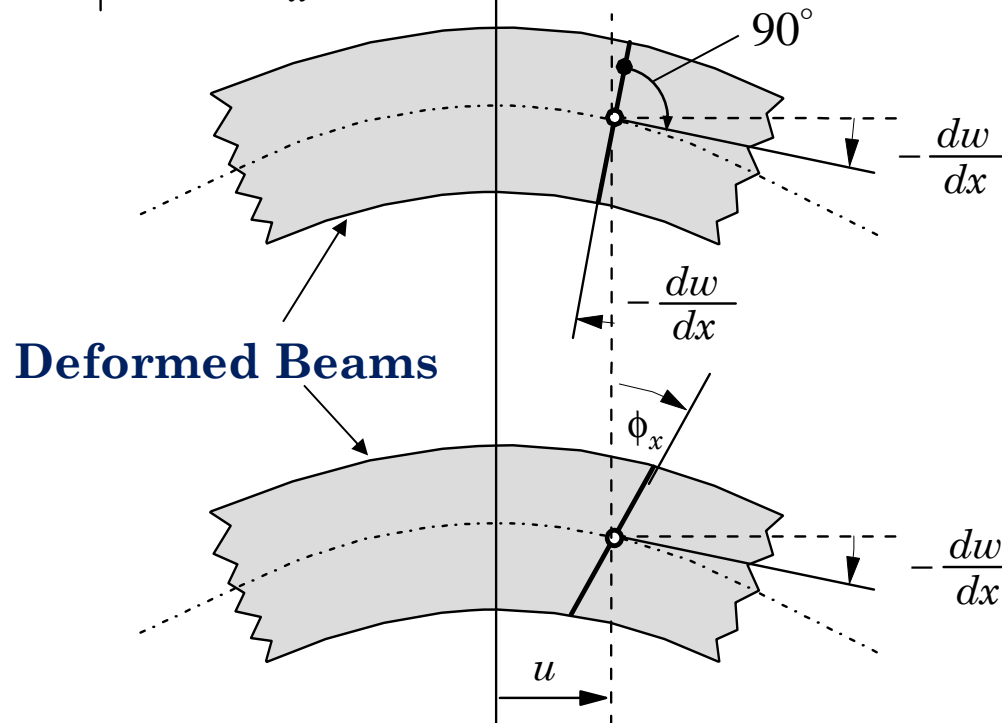
TIMOSHENKO BEAM THEORY and its Finite Element Model

- **Governing Equations**
- **Finite element model**
- **Shear locking**
- **Numerical example**

Kinematics of Timoshenko Beam Theory



Undeformed Beam



Deformed Beams

**Euler-Bernoulli
Beam Theory (EBT)**
*Straightness,
inextensibility, and
normality*

**Timoshenko Beam
Theory (TBT)**
*Straightness and
inextensibility*

Timoshenko Beam Theory

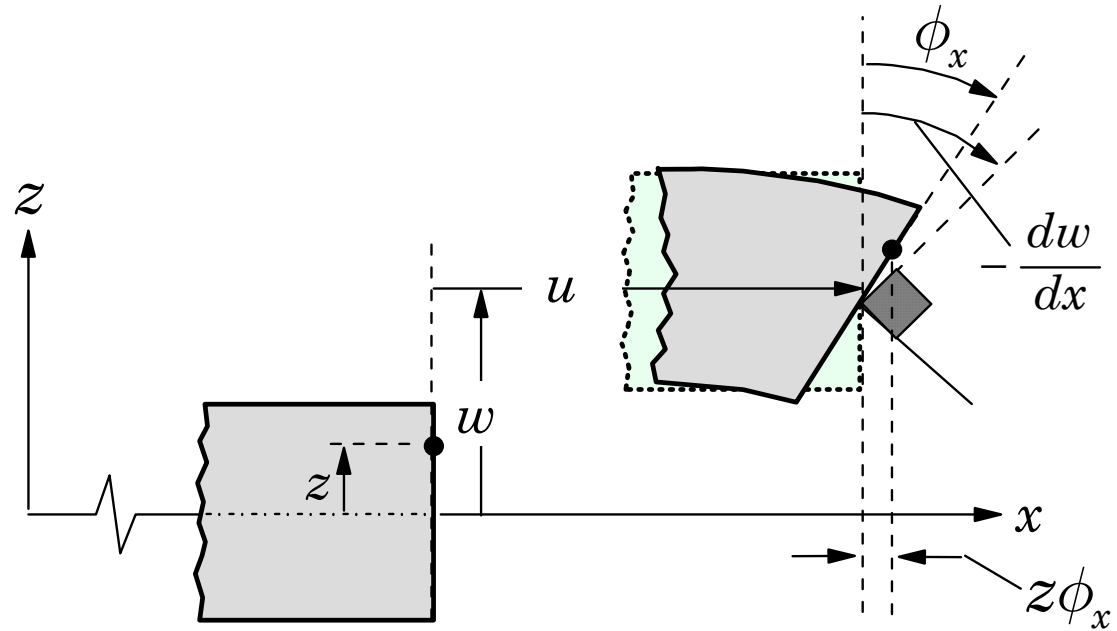
Kinematic Relations

$$u_1(x, z) = u(x) + z\phi_x(x),$$

$$u_2 = 0, \quad u_3(x, z) = w(x)$$

$$\varepsilon_{xx} = \frac{\partial u_1}{\partial x_1} = \frac{du}{dx} + z \frac{d\phi_x}{dx},$$

$$\gamma_{xz} = \frac{\partial u_1}{\partial x_3} + \frac{\partial u_3}{\partial x_1} = \phi_x + \frac{dw}{dx}$$



Constitutive Equations

$$\sigma_{xx} = E \varepsilon_{xx} = E \left(\frac{du}{dx} + z \frac{d\phi_x}{dx} \right)$$

$$\sigma_{xz} = G \gamma_{xz} = G \left(\phi_x + \frac{dw}{dx} \right)$$

Timoshenko Beam Theory (Continued)

Equilibrium Equations $\frac{dN}{dx} + f = 0$, $-\frac{dV}{dx} - q + c_f w = 0$, $-\frac{dM}{dx} + V = 0$.

Beam Constitutive Equations

$$N = \int_A \sigma_{xx} dA = \int_A E \left(\frac{du}{dx} + z \frac{d\phi_x}{dx} \right) dA = EA \frac{du}{dx}$$

$$M = \int_A \sigma_{xx} z dA = \int_A E \left(\frac{du}{dx} + z \frac{d\phi}{dx} \right) z dA = EI \frac{d\phi}{dx}$$

$$V = K_s \int_A \sigma_{xz} dA = GK_s \left(\phi + \frac{dw}{dx} \right) \int_A dA = GAK_s \left(\phi + \frac{dw}{dx} \right)$$

Governing Equations in terms of the displacements

$$-\frac{d}{dx} \left[GAK_s \left(\phi + \frac{dw}{dx} \right) \right] + c_f w = q \quad (1)$$

$$-\frac{d}{dx} \left(EI \frac{d\phi}{dx} \right) + GAK_s \left(\phi + \frac{dw}{dx} \right) = 0 \quad (2)$$

WEAK FORMS OF TBT

Weak Form of Eq. (1)

$$\begin{aligned}
 0 &= \int_{x_a}^{x_b} v_1 \left\{ -\frac{d}{dx} \left[GAK_s \left(\phi + \frac{dw}{dx} \right) \right] + c_f w - q \right\} dx \\
 &= \int_{x_a}^{x_b} \left\{ \frac{dv_1}{dx} \left[GAK_s \left(\phi + \frac{dw}{dx} \right) \right] + c_f v_1 w - v_1 q \right\} dx - \left[v_1 \cdot GAK_s \left(\phi + \frac{dw}{dx} \right) \right]_{x_a}^{x_b} \\
 &= \int_{x_a}^{x_b} \left\{ \frac{dv_1}{dx} \left[GAK_s \left(\phi + \frac{dw}{dx} \right) \right] + c_f v_1 w - v_1 q \right\} dx \\
 &\quad - v_1(x_a) \cdot \left[-GAK_s \left(\phi + \frac{dw}{dx} \right) \right]_{x_a} - v_1(x_b) \cdot \left[GAK_s \left(\phi + \frac{dw}{dx} \right) \right]_{x_b} \\
 &= \int_{x_a}^{x_b} \left\{ GAK_s \frac{dv_1}{dx} \left(\phi + \frac{dw}{dx} \right) + c_f v_1 w - v_1 q \right\} dx - v_1(x_a) \cdot Q_1 - v_1(x_b) \cdot Q_3
 \end{aligned}$$

Weak Forms of TBT (continued)

Weak Form of Eq. (2)

$$\begin{aligned} 0 &= \int_{x_a}^{x_b} v_2 \left[-\frac{d}{dx} \left(EI \frac{d\phi}{dx} \right) + GAK_s \left(\phi + \frac{dw}{dx} \right) \right] dx \\ &= \int_{x_a}^{x_b} \left[\frac{dv_2}{dx} \left(EI \frac{d\phi}{dx} \right) + GAK_s v_2 \left(\phi + \frac{dw}{dx} \right) \right] dx - \left[v_2 \cdot EI \frac{d\phi}{dx} \right]_{x_a}^{x_b} \\ &= \int_{x_a}^{x_b} \left[\frac{dv_2}{dx} \left(EI \frac{d\phi}{dx} \right) + GAK_s v_2 \left(\phi + \frac{dw}{dx} \right) \right] dx - v_2(x_a) \cdot \left(-EI \frac{d\phi}{dx} \right)_{x_a} - v_2(x_b) \cdot \left(EI \frac{d\phi}{dx} \right)_{x_b} \end{aligned}$$

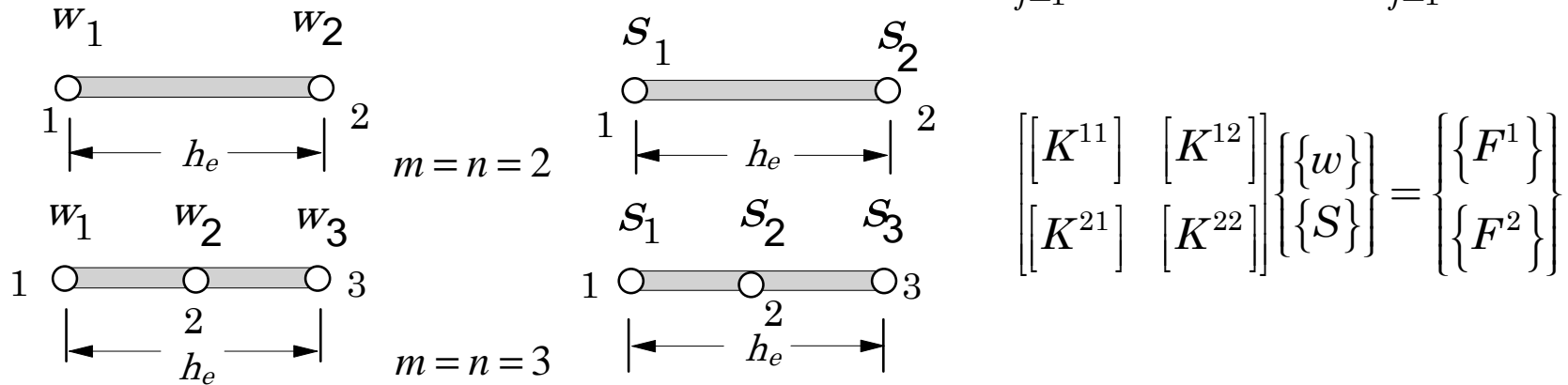
$$0 = \int_{x_a}^{x_b} \left[\frac{dv_2}{dx} \left(EI \frac{d\phi}{dx} \right) + GAK_s v_2 \left(\phi + \frac{dw}{dx} \right) \right] dx - v_2(x_a) \cdot Q_2 - v_2(x_b) \cdot Q_4$$

Total Potential Energy

$$\begin{aligned} \Pi(w, \phi_x) &= \int_{x_a}^{x_b} \left[\frac{EI}{2} \left(\frac{d\phi}{dx} \right)^2 + \frac{GAK_s}{2} \left(\phi + \frac{dw}{dx} \right)^2 + \frac{c_f}{2} w^2 \right] dx \\ &\quad - \int_{x_a}^{x_b} w q dx + w(x_a) Q_1 + w(x_b) Q_3 + \phi(x_a) Q_2 + \phi(x_b) Q_4 \end{aligned}$$

FINITE ELEMENT MODELS OF TIMOSHENKO BEAMS

Finite Element Approximation $w \approx \sum_{j=1}^m w_j \psi_j(x)$, $\phi \approx \sum_{j=1}^n S_j \varphi_j(x)$



$$\begin{bmatrix} [K^{11}] & [K^{12}] \\ [K^{21}] & [K^{22}] \end{bmatrix} \begin{Bmatrix} \{w\} \\ \{S\} \end{Bmatrix} = \begin{Bmatrix} \{F^1\} \\ \{F^2\} \end{Bmatrix}$$

$$K_{ij}^{11} = \int_{x_a}^{x_b} \left(GAK_s \frac{d\psi_i}{dx} \frac{d\psi_j}{dx} + c_f \psi_i \psi_j \right) dx, \quad K_{ij}^{12} = \int_{x_a}^{x_b} GAK_s \frac{d\psi_i}{dx} \varphi_j dx = K_{ji}^{21}$$

$$K_{ij}^{22} = \int_{x_a}^{x_b} \left[EI \frac{d\varphi_i}{dx} \frac{d\varphi_j}{dx} + GAK_s \varphi_i \varphi_j \right] dx, \quad K_{ij}^{21} = \int_{x_a}^{x_b} GAK_s \varphi_i \frac{d\psi_j}{dx} dx$$

$$F_i^1 = \int_{x_a}^{x_b} q \psi_i dx + \psi_i(x_a) Q_1 + \psi_i(x_b) Q_3, \quad F_i^2 = \varphi_i(x_a) Q_2 + \varphi_i(x_b) Q_4$$

Shear Locking in Timoshenko Beams

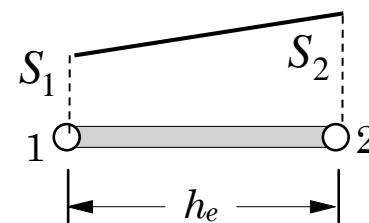
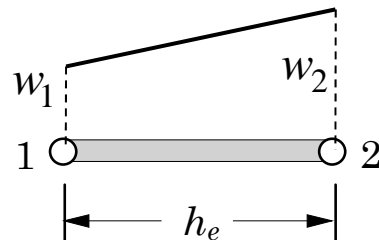
(1) Thick beam experiences shear deformation, $\phi_x \neq -\frac{dw}{dx}$

(2) Shear deformation is negligible in thin beams, $\phi_x = -\frac{dw}{dx}$

Linear interpolation of both w, ϕ_x :

$$w \approx \sum_{j=1}^2 w_j^e \psi_j^e(x), \quad \phi_x \approx \sum_{j=1}^2 S_j^e \psi_j^e(x)$$

$$w(x) \approx w_1 \psi_1(x) + w_2 \psi_2(x), \quad \phi_x(x) \approx S_1 \psi_1(x) + S_2 \psi_2(x)$$



Thus, in the **thin beam limit** it is not possible for the element to realize the requirement

$$\phi_x = -\frac{dw}{dx}$$



SHEAR LOCKING - REMEDY

In the thin beam limit, ϕ should become constant so that it matches dw/dx . However, if ϕ is a constant then the bending energy becomes zero. If we can mimic the two states (constant and linear) in the formulation, we can overcome the problem. Numerical integration of the coefficients allows us to evaluate both ϕ and $d\phi/dx$ as constants. The terms highlighted should be evaluated using “reduced integration”.

$$K_{ij}^{11} = \int_{x_a}^{x_b} \left(GAK_s \frac{d\psi_i^{(1)}}{dx} \frac{d\psi_j^{(1)}}{dx} + c_f \psi_i^{(1)} \psi_j^{(1)} \right) dx$$

$$K_{ij}^{12} = \int_{x_a}^{x_b} GAK_s \frac{d\psi_i^{(1)}}{dx} \psi_j^{(2)} dx = K_{ji}^{21}$$

$$K_{ij}^{22} = \int_{x_a}^{x_b} \left[EI \frac{d\psi_i^{(2)}}{dx} \frac{d\psi_j^{(2)}}{dx} + GAK_s \psi_i^{(2)} \psi_j^{(2)} \right] dx$$

STIFFNESS MATRICES OF TIMOSHENKO BEAM ELEMENT

(for constant EI and GA)

Reduced integration linear element (RIE)

Linear approximation of both w and ϕ

$$\frac{2E_e I_e}{\mu_0^e h_e^3} \begin{bmatrix} 6 & -3h_e & -6 & -3h_e \\ -3h_e & h_e^2 \xi_e & 3h_e & h_e^2 \zeta_e \\ -6 & 3h_e & 6 & 3h_e \\ -3h_e & h_e^2 \zeta_e & 3h_e & h_e^2 \xi_e \end{bmatrix} \begin{Bmatrix} w_1 \\ \phi_1 \\ w_2 \\ \phi_2 \end{Bmatrix} = \begin{Bmatrix} q_1^e \\ q_2^e \\ q_3^e \\ q_4^e \end{Bmatrix} + \begin{Bmatrix} Q_1^e \\ Q_2^e \\ Q_3^e \\ Q_4^e \end{Bmatrix}$$

$$\xi_e = 1.5 + 6\Lambda_e, \quad \zeta_e = 1.5 - 6\Lambda_e, \quad \Lambda_e = \frac{E_e I_e}{G_e A_e K_s h_e^2}, \quad \mu_0^e = 12\Lambda_e$$

Consistent interelement element (CIE)

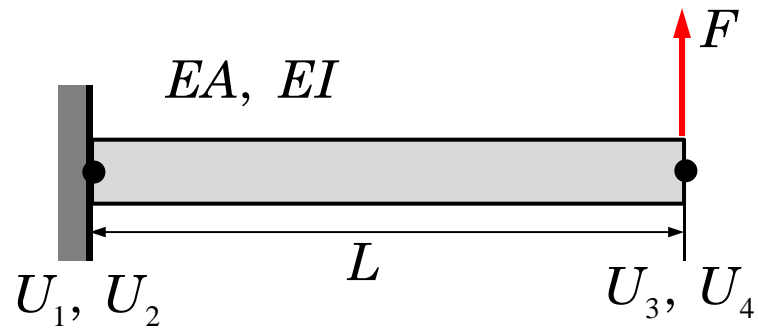
Hermite cubic approximation

of w and dependent quadratic approximation of ϕ

$$\frac{2E_e I_e}{\mu_e h_e^3} \begin{bmatrix} 6 & -3h_e & -6 & -3h_e \\ -3h_e & 2h_e^2 \Sigma_e & 3h_e & h_e^2 \Theta_e \\ -6 & 3h_e & 6 & 3h_e \\ -3h_e & h_e^2 \Theta_e & 3h_e & 2h_e^2 \Sigma_e \end{bmatrix} \begin{Bmatrix} w_1 \\ \phi_1 \\ w_2 \\ \phi_2 \end{Bmatrix} = \begin{Bmatrix} q_1^e \\ q_2^e \\ q_3^e \\ q_4^e \end{Bmatrix} + \begin{Bmatrix} Q_1^e \\ Q_2^e \\ Q_3^e \\ Q_4^e \end{Bmatrix}$$

$$\Sigma_e = 1.0 + 3\Lambda_e, \quad \Theta_e = 1.0 - 6\Lambda_e, \quad \Lambda_e = \frac{E_e I_e}{G_e A_e K_s h_e^2}, \quad \mu_e = 1 + 12\Lambda_e$$

AN EXAMPLE of TBT



Exact solution (according to the E-B beam theory)

$$w(L) = \frac{FL^3}{3EI}$$

One element discretization using the RIE element

$$\frac{2E_e I_e}{\mu_0^e h_e^3} \begin{bmatrix} 6 & -3h_e & -6 & -3h_e \\ -3h_e & h_e^2 \xi_e & 3h_e & h_e^2 \zeta_e \\ -6 & 3h_e & 6 & 3h_e \\ -3h_e & h_e^2 \zeta_e & 3h_e & h_e^2 \xi_e \end{bmatrix} \begin{Bmatrix} w_1 \\ \phi_1 \\ w_2 \\ \phi_2 \end{Bmatrix} = \begin{Bmatrix} q_1^e \\ q_2^e \\ q_3^e \\ q_4^e \end{Bmatrix} + \begin{Bmatrix} Q_1^e \\ Q_2^e \\ Q_3^e \\ Q_4^e \end{Bmatrix}$$

$$\xi_e = 1.5 + 6\Lambda_e, \quad \zeta_e = 1.5 - 6\Lambda_e, \quad \Lambda_e = \frac{E_e I_e}{G_e A_e K_s h_e^2}, \quad \mu_0^e = 12\Lambda_e$$

Boundary conditions:

$$U_1 = U_2 = 0, \quad Q_3 = F, \quad Q_4 = 0$$



AN EXAMPLE (TBT) (continued)

$$\frac{2EI}{\mu_0 L^3} \begin{bmatrix} 6 & 3L \\ 3L & \xi L^2 \end{bmatrix} \begin{Bmatrix} U_3 \\ U_4 \end{Bmatrix} = \begin{Bmatrix} F \\ 0 \end{Bmatrix} \Rightarrow U_3 = \frac{\mu_0 L^3}{2EI} \frac{FL^2 \xi}{(6L^2 \xi - 9L^2)} = \frac{12\Lambda FL^3 (1.5 + 6\Lambda)}{6EI(12\Lambda)}$$

$$\text{When } \Lambda = \frac{EI}{GAK_s L^2} = 0 \Rightarrow U_3 = \frac{1.5FL^3}{6EI} = \frac{FL^3}{4EI} \text{ (too stiff)}$$

$$\text{When } \Lambda \neq 0, \text{ then } U_3 = \frac{FL^3(1.5 + 6\Lambda)}{6EI} = (0.75 + 3\Lambda) \frac{FL^3}{3EI}$$

$$\Lambda = \frac{EI}{GAK_s L^2} = \frac{2(1+\nu)H^2}{12L^2 K_s} = \frac{(1+\nu)}{6K_s} \left(\frac{H}{L}\right)^2 = \frac{1.3}{5} \left(\frac{H}{L}\right)^2 = 0.26 \left(\frac{H}{L}\right)^2$$



AN EXAMPLE of TBT

One element discretization using the CIE element

$$\frac{2EI}{\mu L^3} \begin{bmatrix} 6 & -3L & -6 & -3L \\ -3L & 2L^2\Sigma & 3L & L^2\Theta \\ -6 & 3L & 6 & 3L \\ -3L & L^2\Theta & 3L & 2L^2\Sigma \end{bmatrix} \begin{Bmatrix} w_1 \\ \phi_1 \\ w_2 \\ \phi_2 \end{Bmatrix} = \begin{Bmatrix} Q_1 \\ Q_2 \\ Q_3 \\ Q_4 \end{Bmatrix}$$

$$\Sigma = 1.0 + 3\Lambda, \quad \Theta = 1.0 - 6\Lambda, \quad \Lambda = \frac{EI}{GA K_s L^2}, \quad \mu = 1 + 12\Lambda$$

Condensed equations for the unknown displacements

$$\frac{2EI}{\mu L^3} \begin{bmatrix} 6 & 3L \\ 3L & 2\Sigma L^2 \end{bmatrix} \begin{Bmatrix} U_3 \\ U_4 \end{Bmatrix} = \begin{Bmatrix} F \\ 0 \end{Bmatrix} \Rightarrow U_3 = \frac{\mu L^3}{2EI} \frac{2FL^2\Sigma}{(12L^2\Sigma - 9L^2)} = \frac{\mu FL^3\Sigma}{EI(12\Sigma - 9)}$$



AN EXAMPLE (TBT) (continued)

When $\Lambda = 0 \Rightarrow \Sigma = 1$ and $\mu = 1$; then $U_3 = \frac{\mu FL^3 \Sigma}{EI(12\Sigma - 9)} = \frac{FL^3}{3EI}$

When $\Lambda \neq 0$, $U_3 = \frac{\mu FL^3 \Sigma}{EI(12\Sigma - 9)} = \frac{FL^3}{3EI} \frac{(1 + 3\Lambda)(1 + 12\Lambda)}{(1 + 12\Lambda)} = (1 + 3\Lambda) \frac{FL^3}{3EI}$

$$\Lambda = \frac{EI}{GAK_s L^2} = \frac{2(1 + \nu)H^2}{12L^2 K_s} = \frac{(1 + \nu)}{6K_s} \left(\frac{H}{L} \right)^2 = \frac{1.3}{5} \left(\frac{H}{L} \right)^2 = 0.26 \left(\frac{H}{L} \right)^2$$



SUMMARY

In this lecture we have covered the following topics:

- **Derived the governing equations of the Euler-Bernoulli beam theory**
- **Derived the governing equations of the Timoshenko beam theory**
- **Developed Weak forms of EBT and TBT**
- **Developed Finite element models of EBT and TBT**
- **Discussed shear locking in Timoshenko beam finite element**
- **Discussed assembly of beam elements**
- **Discussed examples**